

The Boussinesq approximation in rapidly rotating flows

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In the classical formulation of the Boussinesq approximation centrifugal buoyancy effects related to differential rotation, as well as strong vortices in the flow, are neglected. However, these may play an important role in rapidly rotating flows, such as in astrophysical and geophysical applications, and also in turbulent convection. We here provide a straightforward approach resulting in a Boussinesq-type approximation that consistently accounts for centrifugal effects. We further compare our new approach to the classical one in fluid flows confined between two differentially heated and rotating cylinders. The results justify the need of using the proposed approximation in rapidly rotating flows.

1. Introduction

In 1903 Boussinesq observed that: “The variations of density can be ignored except where they are multiplied by the acceleration of gravity in the equation of motion for the vertical component of the velocity vector” (Boussinesq 1903). This simple approximation has had far-reaching impact on many areas of fluid dynamics; it allows us to approximate flows with small density variations as incompressible whilst retaining the leading order effects due to the density variations. Moreover, it is of great importance both analytically and numerically as it eliminates certain waves and acoustic modes which are more challenging to treat. Many fluid dynamics problems have taken advantage of Boussinesq-type approximations, rendering in most cases successful results in good agreement with experiments. However, some problems feature important physics neglected in the classical Boussinesq-type formulation. This is the case of fluids subjected to fast rotation. The traditional approach to account for centrifugal buoyancy is solely based on the angular velocity of the rotating reference frame, namely, the centrifugal term in the Navier-Stokes equations is proportional to Ω^2 and acting in the radial direction. This approximation is easy to apply in problems with a distinguished rotating frame of reference. Nevertheless, when such a frame of reference cannot be uniquely identified, the previous approximation neglects significant centrifugal effects due to differential rotation or strong internal vortic-

ity, which are specially important in fast rotating flows. The increasing interest on these flows because of their industrial (e.g. cyclonic dust collectors or vortex chambers) and scientific (astrophysical and atmospheric turbulence) applications (see Elperin *et al.* 1998) motivates the development of a new approximation, which we here undertake. It is based on the classical Boussinesq approximation but it includes additional physical effects stemming from the advection term in the Navier–Stokes equations. This new formulation allows it to accurately cast rapidly rotating flows with mild variations of density into an incompressible formulation. In section §2, we describe a systematic way to achieve this, and we provide two different and easy to implement ways to account for centrifugal buoyancy effects in rotating problems.

In order to illustrate the effect of different ways to account for centrifugal buoyancy in a Boussinesq-type approximation, we study numerically the linear stability of an axially periodic Taylor–Couette system with a negative radial gradient of temperature. Apart from its intrinsic interest, this setting has been widely used to model both atmospheric (e.g., see Randriamampianina *et al.* 2006) and astrophysical flows (e.g. see Petersen *et al.* 2007) where the fluid reaches high rotational speeds.

Our simulations show that the classical Boussinesq approximation is valid in a wide range of Re numbers. However, for flows with high angular velocities important centrifugal effects arise. Here even the linear behaviour of the problem is significantly different for both approximations, justifying the application of the new approximation to account for centrifugal effects in rapidly rotating flows. The rest of the paper is organised as follows. In section §3 we give a detailed description of the system as well as the governing equations of the problem and its linearisation. A brief description of the basic flow is also provided. Section §4 introduces the Petrov–Galerkin method implemented to discretize the equations. In §5 the linear stability of the system considering both ways to introduce the centrifugal buoyancy is compared. Various cases of interest are analysed. In §5.1 we consider fluid rotating as a solid body, whereas in §5.2 shear is introduced in the system. We study quasi-Keplerian rotation in §5.2.1 and a system rotating close to solid body subjected to weak shear in §5.2.2. Discussion and concluding remarks are given in section §6.

2. Boussinesq-type approximation for the centrifugal term

In rotating thermal convection or stratified fluids the Navier–Stokes–Boussinesq equations are usually formulated in the rotating reference frame, with angular velocity vector $\mathbf{\Omega}$. The momentum equation in this non-inertial reference frame includes four inertial body force terms (Batchelor 1967), also called d’Alambert forces:

$$\begin{aligned} \rho(\partial_t + \mathbf{u} \cdot \nabla)\mathbf{u} = & -\nabla p + \nabla \cdot \boldsymbol{\sigma} + \rho \mathbf{f} - \rho \nabla \Phi \\ & - \rho \mathbf{A} - \rho \boldsymbol{\alpha} \times \mathbf{r} - 2\rho \mathbf{\Omega} \times \mathbf{u} - \rho \mathbf{\Omega} \times (\mathbf{\Omega} \times \mathbf{r}). \end{aligned} \quad (2.1)$$

Here $-\rho \mathbf{A}$ is the translation force due to the acceleration \mathbf{A} of the origin of the rotating reference frame, $-\rho \boldsymbol{\alpha} \times \mathbf{r}$ is the azimuthal force (also called Euler force) due to the

angular acceleration $\boldsymbol{\alpha} = d\boldsymbol{\Omega}/dt$, $-2\rho\mathbf{u} \times \boldsymbol{\Omega}$ is the Coriolis force and $-\rho\boldsymbol{\Omega} \times (\boldsymbol{\Omega} \times \mathbf{r})$ is the centrifugal force (all of them per unit volume). In (2.1), ρ , p and \mathbf{u} are the density, pressure and velocity field of the fluid, \mathbf{r} is the position vector of the fluid parcel, and Φ is the gravitational potential, so $-\rho\nabla\Phi$ is the gravitational force. The term $\rho\mathbf{f}$ accounts for additional body forces that may act on the fluid. For a Newtonian fluid the stress tensor $\boldsymbol{\sigma}$ reads

$$\boldsymbol{\sigma} = -p\mathbb{I} + \mu(\nabla\mathbf{u} + \nabla\mathbf{u}^T) + \lambda\nabla \cdot \mathbf{u}\mathbb{I}, \quad (2.2)$$

where \mathbb{I} is the identity tensor, μ is the dynamic viscosity, and λ is the second viscosity.

2.1. The Boussinesq approximation in a rotating reference frame

In most applications (geophysical flows, laboratory experiments), $\boldsymbol{\Omega}$ is constant and $\mathbf{A} = 0$, so we shall here focus on the gravitational, Coriolis and centrifugal terms. Note that the remaining inertial forces could be treated analogously. In the Boussinesq approximation all fluid properties are treated as constant, except for the density, whose variations are considered only in the “relevant” terms. Density variations are assumed to be small: $\rho = \rho_0 + \rho'$, with ρ_0 constant and $\rho'/\rho_0 \ll 1$; the ρ' term usually includes the temperature dependence, density variations due to fluid density stratification, density variations in a binary fluid with miscible species of different densities, etc. With this assumption the continuity equation reduces to $\nabla \cdot \mathbf{u} = 0$ and the fluid can be treated as incompressible. As a direct consequence the shear stress term in the momentum equation (2.1) simplifies to the vector Laplacian, i.e. $\nabla \cdot \boldsymbol{\sigma} = \mu\nabla^2\mathbf{u}$.

Identifying the relevant terms in the momentum equation is a more delicate issue. Any term in (2.1) with a factor ρ splits into two terms, one with a factor ρ_0 and the other with a factor ρ' . If a ρ_0 term is not a gradient, it is the leading-order term, and the associated ρ' term may be neglected. If the ρ_0 term is a gradient, it can be absorbed into the pressure gradient and does not play any dynamical role, and therefore the associated ρ' term must be retained in order to account for the associated force at leading order. This is exactly what happens with the gravitational term: $-\rho_0\nabla\Phi = \nabla(-\rho_0\Phi)$, which is absorbed into the pressure gradient term and we must retain the $-\rho'\nabla\Phi$ term to account for gravitational buoyancy. The same treatment must be applied to the translation and centrifugal terms, yielding the gradient terms

$$-\rho_0\mathbf{A} - \rho_0\boldsymbol{\Omega} \times (\boldsymbol{\Omega} \times \mathbf{r}) = \nabla\left(\frac{1}{2}\rho_0|\boldsymbol{\Omega} \times \mathbf{r}|^2 - \rho_0\mathbf{A} \cdot \mathbf{r}\right), \quad (2.3)$$

as well as $-\rho'\mathbf{A}$ and $-\rho'\boldsymbol{\Omega} \times (\boldsymbol{\Omega} \times \mathbf{r})$, which must be also retained.

The ρ_0 part of the remaining terms in equation (2.1) (so far, we have considered the gravitational, centrifugal and translational forces) are not gradients, so they are retained as leading order terms and the corresponding ρ' terms are neglected, leading to the Boussinesq approximation equations in the rotating reference frame:

$$\begin{aligned} \rho_0(\partial_t + \mathbf{u} \cdot \nabla)\mathbf{u} &= -\nabla p^* + \mu\nabla^2\mathbf{u} + \rho\mathbf{f} - \rho'\nabla\Phi \\ &\quad - \rho'\mathbf{A} - \rho_0\boldsymbol{\alpha} \times \mathbf{r} - 2\rho_0\boldsymbol{\Omega} \times \mathbf{u} - \rho'\boldsymbol{\Omega} \times (\boldsymbol{\Omega} \times \mathbf{r}), \end{aligned} \quad (2.4)$$

where

$$p^* = p + \rho_0 \Phi - \frac{1}{2} \rho_0 |\boldsymbol{\Omega} \times \mathbf{r}|^2 + \rho_0 \mathbf{A} \cdot \mathbf{r}, \quad (2.5)$$

together with the incompressibility condition $\nabla \cdot \mathbf{u} = 0$. Of course, supplementary equations are often needed; for example, if ρ' depends on the temperature, an evolution equation for the temperature must be included.

2.2. Formulation in the inertial frame

In many cases the fluid container is not rotating at a given angular speed, but different parts may rotate independently. For example Taylor-Couette flows with stratification and/or heating, cylindrical containers with the lids rotating at different angular velocities, etc. In these flows, there is not a natural or unique angular velocity $\boldsymbol{\Omega}$ to use in (2.4) and it may be more convenient to write the governing equations in the laboratory reference frame. In §2.2.1 we derive the momentum equation in the laboratory frame but for the sake of simplicity we assume that the fluid container rotates with angular speed $\boldsymbol{\Omega}$. In §2.2.2 we show how the formulation is easily extended to account for the general case where a unique rotating reference frame cannot be identified.

2.2.1. Formulation in the inertial frame: container rotating at angular velocity $\boldsymbol{\Omega}$

The laboratory frame is an inertial reference frame, so the four inertial terms in (2.1) are absent, and the balance momentum equation is

$$\rho(\partial_t + \mathbf{v} \cdot \nabla) \mathbf{v} = -\nabla p + \mu \nabla^2 \mathbf{v} - \rho \nabla \Phi + \rho \mathbf{f}, \quad (2.6)$$

where we have used \mathbf{v} for the velocity field in the inertial reference frame, to distinguish it from the velocity in the rotating frame \mathbf{u} . In order to implement the Boussinesq approximation, we could naïvely repeat the previous analysis; since the only term which is a gradient is the gravitational force $-\rho_0 \nabla \Phi$, we end up with an equation containing only the gravitational buoyancy, and the centrifugal buoyancy is absent. This appears reasonable, because the governing equations do not contain the rotation frequency $\boldsymbol{\Omega}$ of the container. However, $\boldsymbol{\Omega}$ appears in the boundary conditions for the velocity, so it must be taken into account by a careful analysis of the nonlinear advection term. The easiest way to do this is by decomposing the velocity field as $\mathbf{v} = \mathbf{u} + \boldsymbol{\Omega} \times \mathbf{r}$, so the $\boldsymbol{\Omega} \times \mathbf{r}$ part accounts for the boundary conditions (rotating container); \mathbf{u} is precisely the velocity of the fluid in the rotating reference frame, with zero velocity boundary conditions. The advection term splits into four parts:

$$\mathbf{v} \cdot \nabla \mathbf{v} = \mathbf{u} \cdot \nabla \mathbf{u} + \mathbf{u} \cdot \nabla (\boldsymbol{\Omega} \times \mathbf{r}) + (\boldsymbol{\Omega} \times \mathbf{r}) \cdot \nabla \mathbf{u} + (\boldsymbol{\Omega} \times \mathbf{r}) \cdot \nabla (\boldsymbol{\Omega} \times \mathbf{r}). \quad (2.7)$$

Using the incompressibility character of \mathbf{u} , the dependence of $\boldsymbol{\Omega}$ on time but not on the spatial coordinates, and some vector identities, we can transform the advection term into

$$\mathbf{v} \cdot \nabla \mathbf{v} = \mathbf{u} \cdot \nabla \mathbf{u} + 2 \boldsymbol{\Omega} \times \mathbf{u} + \boldsymbol{\Omega} \times (\boldsymbol{\Omega} \times \mathbf{r}) + \nabla \times (\mathbf{u} \times (\boldsymbol{\Omega} \times \mathbf{r})). \quad (2.8)$$

We have recovered the Coriolis and centrifugal terms, and because $\boldsymbol{\Omega} \times (\boldsymbol{\Omega} \times \mathbf{r})$ is a gradient, we must add a centrifugal contribution also in the inertial reference frame.

The last term in (2.8) accounts for the difference between the time derivatives in the inertial and rotating reference frames respectively. An easy way to see this is by considering the simple case where the two reference frames have the same origin, and $\mathbf{\Omega} = \Omega \hat{\mathbf{k}}$, where $\hat{\mathbf{k}}$ is the vertical unit vector and Ω is constant. Using cylindrical coordinates (r, θ, z) , with z in the vertical direction, we obtain

$$\nabla \times (\mathbf{u} \times (\mathbf{\Omega} \times \mathbf{r})) = \Omega \partial_\theta \mathbf{u}. \quad (2.9)$$

The change of coordinates between the inertial and rotating frame is

$$\left. \begin{aligned} r &= r', & z &= z', \\ \theta &= \theta' + \Omega t, & t &= t', \end{aligned} \right\} \quad (2.10)$$

where (r', θ', z') are the cylindrical coordinates in the rotating frame of the same fluid parcel with coordinates (r, θ, z) in the inertial frame; t and t' are the times in both reference frames. From (2.10) we obtain $\partial_{t'} = \partial_t + \Omega \partial_\theta$, so the last term in (2.8), combined with $\partial_t \mathbf{u}$ results in the term $\partial_{t'} \mathbf{u}$ in the rotating frame. Finally, $\partial_t \mathbf{v}$ in the inertial frame contains an extra term, $\partial_t (\mathbf{\Omega} \times \mathbf{r}) = \mathbf{\alpha} \times \mathbf{r}$. Therefore, we have recovered all the inertial forces in the rotating frame (2.1), except for the translation force $-\rho \mathbf{A}$, because in the example considered, (2.10), both reference frames have the same origin, and the translation is absent; by including a translation term in (2.10) we could also recover it. Now, the two formulations, including centrifugal buoyancy in both reference frames (inertial and rotating), fully agree.

In the inertial reference frame, we are interested in a formulation in terms of the velocity field in the inertial frame \mathbf{v} , instead of \mathbf{u} as in (2.8). The analysis presented above considering the advection term results simply in an additional term, the centrifugal buoyancy. We have also discussed the effect of the decomposition $\mathbf{v} = \mathbf{u} + \mathbf{\Omega} \times \mathbf{r}$ in the time derivative term. Now it only remains to consider the viscous term. However, $\nabla^2 (\mathbf{\Omega} \times \mathbf{r}) = 0$ because $\mathbf{\Omega} \times \mathbf{r}$ is linear in the spatial coordinates and so its Laplacian is zero. The traditional Boussinesq approximation equations in the inertial reference frame are

$$\rho_0 (\partial_t + \mathbf{v} \cdot \nabla) \mathbf{v} = -\nabla p^* + \mu \nabla^2 \mathbf{v} + \rho \mathbf{f} - \rho' \nabla \Phi - \rho' \mathbf{\Omega} \times (\mathbf{\Omega} \times \mathbf{r}), \quad (2.11)$$

where $p^* = p + \rho_0 \Phi - \frac{1}{2} \rho_0 |\mathbf{\Omega} \times \mathbf{r}|^2$, and together with the incompressibility condition $\nabla \cdot \mathbf{u} = 0$.

2.2.2. Formulation in the inertial frame: generalization

We have shown that centrifugal buoyancy enters the governing equations via the boundary conditions and the advection term; no other term is affected in the Boussinesq approximation. This now suggests a very simple formulation, consisting in keeping the whole density, $\rho = \rho_0 + \rho'$, in the advection term. This formulation is easy to implement, and since most time-evolution codes for incompressible flows are semi-implicit (i.e. the viscous term is treated implicitly, whereas the advection term is treated explicitly), the

speed and efficiency of the codes do not change. The formulation reads

$$\rho_0(\partial_t + \mathbf{v} \cdot \nabla)\mathbf{v} = -\nabla p^* + \mu \nabla^2 \mathbf{v} + \rho \mathbf{f} - \rho' \nabla \Phi - \rho'(\mathbf{v} \cdot \nabla)\mathbf{v}, \quad (2.12)$$

where $p^* = p + \rho_0 \Phi$, and allows one to easily handle situations where different parts of a fluid container rotate independently. In these flows there is not a natural or unique angular velocity $\boldsymbol{\Omega}$ to use for a rotating reference frame in the formulation (2.11); however the angular velocities of the problem still enter the governing equations through the boundary conditions and the advection term. Hence formulation (2.12) provides a natural way to account for centrifugal buoyancy effects of these rotating flows in the inertial (laboratory) reference frame. This formulation is also appropriate if additional equations appear coupled with the Navier-Stokes equations, for example for large density variations in stratified flows. The treatment of the centrifugal effects can be carried out exactly in the same way presented here.

2.2.3. Alternative formulation in the inertial frame and physical interpretation

The extra term included in (2.12), $\rho'(\mathbf{v} \cdot \nabla)\mathbf{v}$, can be expressed in a different way, providing a closer resemblance to the expression in (2.11). Close to a rotating wall, the velocity field is $\mathbf{v} \approx \boldsymbol{\Omega} \times \mathbf{r}$; this expression is exact at the wall (no slip boundary condition at a rigid rotating wall). The dominant part of the advection term is then

$$(\mathbf{v} \cdot \nabla)\mathbf{v} \approx (\boldsymbol{\Omega} \times \mathbf{r}) \cdot \nabla(\boldsymbol{\Omega} \times \mathbf{r}) = \boldsymbol{\Omega} \times (\boldsymbol{\Omega} \times \mathbf{r}) = -\nabla\left(\frac{1}{2}|\boldsymbol{\Omega} \times \mathbf{r}|^2\right) \approx -\nabla\left(\frac{1}{2}\mathbf{v}^2\right). \quad (2.13)$$

As the dominant term is a gradient, it is necessary to include the ρ' term in the Boussinesq approximation. Replacing $\rho'(\mathbf{v} \cdot \nabla)\mathbf{v}$ by $-\rho'\nabla(\frac{1}{2}\mathbf{v}^2)$ gives the alternative form for (2.12):

$$\rho_0(\partial_t + \mathbf{v} \cdot \nabla)\mathbf{v} = -\nabla p^* + \mu \nabla^2 \mathbf{v} + \rho \mathbf{f} - \rho' \nabla \Phi + \rho' \nabla\left(\frac{1}{2}\mathbf{v}^2\right). \quad (2.14)$$

This centrifugal effect is not only important when we have rotating walls, but also if a strong vortex appears dynamically in the interior of the domain; therefore it is advisable to always include this term in the Boussinesq approximation in order to account for all possible sources of centrifugal instability.

We have presented two different ways, (2.12) and (2.14), of including the centrifugal buoyancy in rotating problems. One may wonder if there exists a canonical way to extract from the advection term the part that is a gradient, and then multiply this gradient by ρ' . The Helmholtz decomposition (Arfken & Weber 2005), writing a given vector field as the sum of a gradient and a curl, could serve this purpose, but unfortunately this decomposition is not unique (it depends on the boundary conditions satisfied by the curl part), and moreover it is not a local decomposition (i.e., in order to extract the gradient part, we need to solve a Laplace equation with Neumann boundary conditions). The two formulations presented here, (2.12) and (2.14), are simple and easy to implement, and deciding between one or the other is a matter of taste.

The extra term we have included in (2.14), $\rho'\nabla(\frac{1}{2}\mathbf{v}^2)$, has an important physical interpretation; it is a source of vorticity due to density variations and centrifugal effects.

Taking the curl of (2.14) and using

$$\nabla \times (\mathbf{v} \cdot \nabla \mathbf{v}) = \nabla \times (\boldsymbol{\omega} \times \mathbf{v}) = \mathbf{v} \cdot \nabla \boldsymbol{\omega} - \boldsymbol{\omega} \cdot \nabla \mathbf{v}, \quad (2.15)$$

where $\boldsymbol{\omega} = \nabla \times \mathbf{v}$ is the vorticity field, results in an equation for the vorticity:

$$\rho_0(\partial_t + \mathbf{v} \cdot \nabla) \boldsymbol{\omega} = \rho_0 \boldsymbol{\omega} \cdot \nabla \mathbf{v} + \mu \nabla^2 \boldsymbol{\omega} + \nabla \times (\rho \mathbf{f}) - \nabla \rho' \times \nabla \Phi + \nabla \rho' \times \nabla \left(\frac{1}{2} \mathbf{v}^2 \right). \quad (2.16)$$

The first three terms in the right-hand-side of (2.16) provide the classical vorticity evolution equation for an incompressible flow with constant density. The last two terms are the explicit generation of vorticity due to the gravitational and centrifugal buoyancies, respectively. In the following sections, we compare from the point of view of the linear stability the differences between the classical Boussinesq approach (2.11) and the first formulation proposed in this section, i.e. equation (2.12).

3. Description of the system

3.1. Governing equations

We consider the motion of a fluid of kinematic viscosity ν contained in the annular gap between two concentric cylinders of radii r_i and r_o . The cylinders rotate at independent angular speeds Ω_i and Ω_o . A negative radial gradient of temperature is also considered by setting the temperature of the inner cylinder to $T_i = T_c + \Delta T/2$ and the outer cylinder to $T_o = T_c - \Delta T/2$ where T_c is the mean temperature. We assume periodicity in the axial direction. The centrifugal buoyancy in the stationary frame of reference is included as in §2, equation(2.12):

$$\rho_0(\partial_t + \mathbf{v} \cdot \nabla) \mathbf{v} = -\nabla p^* + \mu \nabla^2 \mathbf{v} - \rho' \nabla \Phi - \rho' \mathbf{v} \cdot \nabla \mathbf{v}, \quad (3.1)$$

where p^* includes part of the gravitational potential, $\rho_0 \Phi$.

We assume $\rho = \rho_0 + \rho' = \rho_0(1 - \alpha T)$, where T is the deviation of the temperature with respect to the mean temperature T_c , and ρ_0 is the density of the fluid at T_c . We assume the gravity acceleration is vertical and uniform, so the gravitational potential is given by $\Phi = gz$; cylindrical coordinates (r, θ, z) are used. With these assumptions, $-\rho' \nabla \Phi = \rho_0 \alpha g T \hat{\mathbf{z}}$ where $\hat{\mathbf{z}}$ is the unit vector in the axial direction z and α is the coefficient of volume expansion. The governing equations, including the temperature and incompressibility condition, are:

$$(\partial_t + \mathbf{v} \cdot \nabla) \mathbf{v} = -\nabla p + \nu \nabla^2 \mathbf{v} + \alpha g T \hat{\mathbf{z}} + \alpha T \mathbf{v} \cdot \nabla \mathbf{v}, \quad (3.2a)$$

$$(\partial_t + \mathbf{v} \cdot \nabla) T = \kappa \nabla^2 T, \quad (3.2b)$$

$$\nabla \cdot \mathbf{v} = 0, \quad (3.2c)$$

where κ is the thermal diffusivity of the fluid. The equations are made dimensionless using the gap width $d = r_o - r_i$ as the length scale, the viscous time d^2/ν as the time scale, ΔT as the temperature scale, and $(\nu/d)^2$ for the pressure. In doing so, six independent

dimensionless numbers appear:

$$\text{Grashof number} \quad G = \alpha g \Delta T d^3 / \nu^2, \quad (3.3a)$$

$$\text{relative density variation} \quad \epsilon = \alpha \Delta T = \Delta \rho / \rho_0, \quad (3.3b)$$

$$\text{Prandtl number} \quad \sigma = \nu / \kappa, \quad (3.3c)$$

$$\text{radius ratio} \quad \eta = r_i / r_o, \quad (3.3d)$$

$$\text{inner Reynolds number} \quad Re_i = \Omega_i r_i d / \nu, \quad (3.3e)$$

$$\text{outer Reynolds number} \quad Re_o = \Omega_o r_o d / \nu. \quad (3.3f)$$

where $\Delta \rho$ is the density variation associated with a temperature change of ΔT . In this system the Froude number is not particularly useful because we have two different rotation rates, Ω_i and Ω_o , so the Froude number definition is not unique.

From now on, only dimensionless variables and parameters will be used. The dimensionless governing equations are:

$$(\partial_t + \mathbf{v} \cdot \nabla) \mathbf{v} = -\nabla p + \nabla^2 \mathbf{v} + GT \hat{\mathbf{z}} + \epsilon T \mathbf{v} \cdot \nabla \mathbf{v}, \quad (3.4a)$$

$$(\partial_t + \mathbf{v} \cdot \nabla) T = \sigma^{-1} \nabla^2 T, \quad (3.4b)$$

$$\nabla \cdot \mathbf{v} = 0, \quad (3.4c)$$

The only change needed to recover the traditional Boussinesq approximation is to replace the centrifugal term $\epsilon T \mathbf{v} \cdot \nabla \mathbf{v}$ in (3.4a) by $-\epsilon \Omega^2 T r \hat{\mathbf{r}}$, where $\hat{\mathbf{r}}$ is the unit vector in the radial direction r .

Due to the large number of parameters governing the system, a comprehensive exploration of the parameter space is extremely expensive, so we will vary Re_i and G while keeping the remaining parameters constant (or a simple function of the two control parameters Re_i and G). In the present study, we keep fixed $\eta = 0.71$, a typical value in experimental facilities, and $\sigma = 7.16$, corresponding to water.

3.2. Basic flow

An analytical solution for the basic flow can be found by assuming only radial dependence for the variables of the problem. We also use the zero axial mass flux condition to fix the axial pressure gradient, i.e.:

$$\int_{r_i}^{r_o} r w_b(r) dr = 0. \quad (3.5)$$

The resulting steady basic flow is given by:

$$u_b(r) = 0 \quad (3.6a)$$

$$v_b(r) = Ar + \frac{B}{r} \quad (3.6b)$$

$$w_b(r) = G \left(C(r^2 - r_i^2) + \left(C(r_o^2 - r_i^2) + \frac{1}{4}(r_o^2 - r^2) \right) \frac{\ln(r/r_i)}{\ln \eta} \right) \quad (3.6c)$$

$$T_b(r) = \frac{1}{2} + \frac{\ln(r/r_i)}{\ln \eta} \quad (3.6d)$$

$$p(r, z) = p_o + G \left(4C + \frac{1}{2} - \frac{1}{\ln \eta} \right) z + \int_{r_i}^r (1 - \epsilon T_b(r)) v_b^2(r) \frac{dr}{r}. \quad (3.6e)$$

where v_b is the azimuthal velocity for the classical Taylor-Couette problem (Chandrasekhar 1961), whereas w_b and T_b correspond to convection in a conductive regime and appeared for the first time in Choi & Korpela (1980). The pressure varies linearly with the axial coordinate z , but the pressure gradient depends only on r , and therefore it is periodic in the axial direction. This axial pressure gradient mimics the presence of far away endwalls in any real situation, by unforcing the zero mass flux constraint (3.5). It is possible to give an explicit closed expression for p by integrating (3.6e), but it is quite involved and it does not appear in the problem solution. The expressions for the parameters A , B and C are:

$$A = \frac{Re_o - \eta Re_i}{1 + \eta}, \quad B = \eta \frac{Re_i - \eta Re_o}{(1 - \eta)(1 - \eta^2)}, \quad (3.7)$$

$$C = -\frac{4 \ln \eta + (1 - \eta^2)(3 - \eta^2)}{16(1 - \eta^2)((1 + \eta^2) \ln \eta + 1 - \eta^2)}, \quad (3.8)$$

where (3.7) define the pure rotational flow in the azimuthal coordinate and C gives the axial component of the velocity field. The non-dimensional radii of the cylindrical walls are given by $r_i = \eta/(1 - \eta)$, $r_o = 1/(1 - \eta)$. Note that the presence of the new centrifugal buoyancy term, proportional to ϵ , does not modify the basic flow's velocity field, but only its pressure.

3.3. Linearized equations

We perturb the basic flow with infinitesimal perturbations which vary periodically in the axial and azimuthal directions,

$$\mathbf{v}(r, \theta, z, t) = \mathbf{v}_b(r) + e^{i(n\theta + kz) + \lambda t} \mathbf{u}(r), \quad (3.9a)$$

$$T(r, \theta, z, t) = T_b(r) + e^{i(n\theta + kz) + \lambda t} T'(r), \quad (3.9b)$$

where $\mathbf{v}_b = (0, v_b, w_b)$ and $T_b(r)$ correspond to the basic flow (3.6); $\mathbf{u}(r) = (u_r, u_\theta, u_z)$ and $T'(r)$ are the velocity and temperature perturbations, respectively. The boundary conditions for both \mathbf{u} and T' are homogeneous: $\mathbf{u}(r_i) = \mathbf{u}(r_o) = T'(r_i) = T'(r_o) = 0$. The axial wavenumber k and the azimuthal mode n define the shape of the disturbance. The parameter λ is complex. Its real part λ_r is the perturbation's growth rate, which

is zero at critical values, and its imaginary part λ_i is the oscillation frequency of the perturbation.

Using the decomposition (3.9) in the equations (3.4) and neglecting high-order terms, we obtain an eigenvalue problem, with eigenvalue λ . It reads

$$\begin{aligned} \lambda u_r = & \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial u_r}{\partial r} \right) - u_r \left[\frac{n^2 + 1}{r^2} + k^2 + i \left(\frac{nv_b}{r} + kw_b \right) (1 - \epsilon T_b) \right] \\ & + \frac{2v_b}{r} (1 - \epsilon T_b) u_\theta - \frac{2in}{r^2} u_\theta - \frac{\epsilon v_b^2}{r} T', \end{aligned} \quad (3.10a)$$

$$\begin{aligned} \lambda u_\theta = & \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial u_\theta}{\partial r} \right) - u_\theta \left[\frac{n^2 + 1}{r^2} + k^2 + i \left(\frac{nv_b}{r} + kw_b \right) (1 - \epsilon T_b) \right] \\ & - \left(\frac{\partial v_b}{\partial r} + \frac{v_b}{r} \right) (1 - \epsilon T_b) u_r + \frac{2in}{r^2} u_r, \end{aligned} \quad (3.10b)$$

$$\begin{aligned} \lambda u_z = & \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial u_z}{\partial r} \right) - u_z \left[\frac{n^2}{r^2} + k^2 + i \left(\frac{nv_b}{r} + kw_b \right) (1 - \epsilon T_b) \right] \\ & + \frac{\partial w_b}{\partial r} (\epsilon T_b - 1) u_z + GT', \end{aligned} \quad (3.10c)$$

$$\lambda T' = \frac{1}{\sigma r} \frac{\partial}{\partial r} \left(r \frac{\partial T'}{\partial r} \right) - T' \left[\frac{1}{\sigma} \left(\frac{n^2}{r^2} + k^2 \right) + i \left(\frac{nv_b}{r} + kw_b \right) \right] - \frac{\partial T_b}{\partial r} u_r. \quad (3.10d)$$

Note that here the continuity equations and pressure terms are omitted because the Petrov–Galerkin method chosen to solve the resulting system of equations automatically satisfies the continuity equation and eliminates the pressure by using a proper projection (see next section).

From (3.10) the equations for the traditional Boussinesq approximation can be easily obtained from (3.10) by setting $\epsilon = 0$ in all terms except for $-\epsilon(v_b^2/r)T'$. The traditional approximation contemplates only one rotating frame of reference for the system; the expression (3.6b) for the basic flow azimuthal velocity $v_b(r) = Ar + B/r$ has two terms, Ar corresponding to solid body rotation, and B/r corresponding to shear. It is natural to identify A as the frequency of the rotating frame of reference, Ω_r . In fact, if we take $\Omega_i = \Omega_o = \Omega$, the Couette flow profile is:

$$v_b(r) = Ar + \frac{B}{r} = \frac{\Omega_o r_o^2 - \Omega_i r_i^2}{r_o^2 - r_i^2} r + \frac{(\Omega_i - \Omega_o)(r_i r_o)^2}{r_o^2 - r_i^2} \frac{1}{r} = \Omega r = \Omega_r r, \quad (3.11)$$

and we recover the linearized version of the centrifugal term considered by the classical approach, $-\epsilon \Omega^2 T' r \hat{\mathbf{r}}$. In the general case with $\Omega_i \neq \Omega_o$ the traditional Boussinesq approximation is defined in the frame of reference rotating with $\Omega_r = A$. This approximation takes only into account the centrifugal buoyancy acting in the radial direction, which is obviously its main contribution. However, as we will see in §5, for high rotation rates other terms acting both in the radial and azimuthal directions become important and change the behaviour of the system. Part of the discrepancy stems from the fact that the effect of differential rotation is entirely neglected in the classical approximation.

4. Numerical method

In order to solve numerically the eigenvalue problem described in the previous section, a spatial discretization of the domain must be made. This is accomplished by projecting the equations (3.10) onto a basis carefully chosen to simplify the process,

$$V_3 = \{\mathbf{v} \in (\mathcal{L}_2(r_i, r_o))^3 \mid \nabla \cdot \mathbf{v} = 0, \mathbf{v}(r_i) = \mathbf{v}(r_o) = 0\}, \quad (4.1)$$

where $(\mathcal{L}_2(r_i, r_o))^3$ is the Hilbert space of square integrable vectorial functions defined on the interval (r_i, r_o) , with the inner product

$$\langle \mathbf{v}, \mathbf{u} \rangle = \int_{r_i}^{r_o} \mathbf{v}^* \cdot \mathbf{u} \, r \, dr, \quad (4.2)$$

where $*$ denotes the complex conjugate. For any $\mathbf{v} \in V_3$ and any function p , we have $\langle \mathbf{v}, \nabla p \rangle = 0$. This consideration allows us to eliminate the pressure from the equations as we project them onto the basis. Moreover, the continuity equation is satisfied by definition of the space V_3 . For the temperature perturbation the appropriate space is

$$V_1 = \{f \in \mathcal{L}_2(r_i, r_o) \mid f(r_i) = f(r_o) = 0\}. \quad (4.3)$$

We expand the variables of the problem as follows

$$\mathbf{X} = \begin{bmatrix} \mathbf{u}(r) \\ T'(r) \end{bmatrix} = \sum_j a_j \mathbf{X}_j \quad \mathbf{X}_j \in V_3 \times V_1, \quad (4.4)$$

and projecting (3.10) onto $V_3 \times V_1$, we arrive at a linear system of equations for the coefficients a_j .

The solution of the system is performed by means of a Petrov-Galerkin scheme, where the basis used in the expansion is different from the one used in the projection. The bases are composed of functions built on Chebyshev polynomials satisfying the boundary conditions. A detailed description of the method as well as the basis and functions used for the velocity field can be found in Meseguer & Marques (2000) and Meseguer *et al.* (2007), respectively. The basis functions for the temperature (last component of \mathbf{X}_j in 4.4), and for the projection (with $\tilde{\cdot}$) are:

$$h_j(r) = (1 - y^2)T_{j-1}(y), \quad \tilde{h}_j(r) = r^2(1 - y^2)T_{j-1}(y), \quad (4.5)$$

where $y = 2(r - r_i) - 1$ and T_j are the Chebyshev polynomials. As a result of this process, we obtain a generalized eigenvalue system of the form

$$\lambda M_1 x = M_2 x, \quad (4.6)$$

where x is a vector containing the complex spectral coefficients (a_j) and the matrices M_1 and M_2 depend on the parameters of the problem, the axial wavenumber k and the azimuthal mode n . This system is solved by using *LAPACK*. The numerical code written to perform this work implements the described method and analyses a range of k , n and G provided by the user for a fixed Re number, searching for the critical values ($\Re \lambda = \lambda_r = 0$). Up to $M = 200$ radial modes have been used in order to ensure the spectral convergence when high Re numbers are considered. The code has been tested by

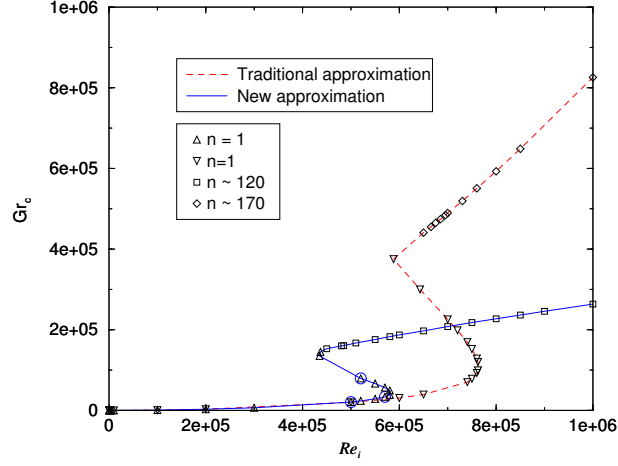


FIGURE 1. Critical Grashof number G_c as function of inner cylinder Reynolds number Re_i for fluid rotating as a solid-body. The solid line is the linear stability curve using the new approximation for the centrifugal buoyancy proposed in this paper, whereas the dashed corresponds to the traditional Boussinesq approach. Different symbols indicate the two distinct mechanisms of instability. Up and down triangles represent the critical points due to the mechanism at moderate Re_i for the new and traditional approximations respectively, whereas squares and diamonds correspond to the mechanism at large Re_i .

computing critical values for several cases in McFadden *et al.* (1984) and Ali & Weidman (1990), obtaining an excellent agreement with their results.

5. Stability of differentially heated fluid between co-rotating cylinders

We present a detailed comparison of the linear stability of the system using the traditional Boussinesq approximation and our new approximation (3.10). We consider three different cases, all with $\eta = 0.71$ and $\sigma = 7.16$. In the first one the cylinders are rotating at same angular speed, corresponding to fluid rotating as a solid-body, and in the second and third cases the stability of a differentially rotating fluid, of importance in astrophysical flows, is considered in the presence of weak and strong shear.

5.1. Cylinders rotating at same angular speed

In this case a rotating frame of reference is readily identified and the shear term B/r in the basic flow azimuthal velocity (3.6b) is zero, whereas the term A corresponds to the angular velocity of the cylinders. Figure 1 shows the critical values of G as the rotation speed, indicated here by the inner cylinder Reynolds number Re_i , is increased. In the case of stationary cylinders instability sets in at $G = 8087.42$, with $k_c = -2.74$ and $n = 0$. The emerging pattern is characterized by pairs of counter-rotating torodial rolls, that unlike Taylor vortices have a non-zero phase velocity that causes a slow drift of the cellular pattern upward. Extensive information about natural convection instabilities can be found in the literature: Choi & Korpela (1980) and McFadden *et al.* (1984) for infinite geometries, and de Vahl Davis & Thomas (1969) and Lee *et al.* (1982) for finite

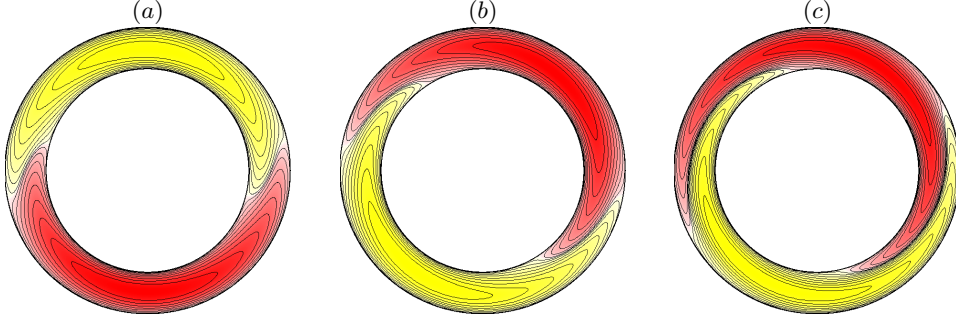


FIGURE 2. Contours of the temperature disturbance T' at a z -constant section corresponding to the points marked as blue circles in figure 1. (a): $Re_i = 5 \cdot 10^5$, $G_c = 21206.53$. (b): $Re_i = 5.7 \cdot 10^5$, $G_c = 33768.37$. (c): $Re_i = 5.2 \cdot 10^5$, $G_c = 79670.16$. There are 10 positive (dark gray; red in the online version) and 10 negative (light gray; yellow in the online version) linearly spaced contours. In all cases the critical azimuthal mode is $n = 1$ and $k = O(10^{-3})$.

geometries. Obviously, without rotation both traditional (dashed line) and new (solid line) yield identical results. As rotation is increased no differences in the linear behaviour of the system are observed up to $Re_i \sim 5 \times 10^5$, where the two curves start to depart from each other. Up to this point and after a small initial region where several azimuthal modes up to $n = 6$ are involved, the basic flow loses stability to an azimuthal mode $n = 1$ with small axial wavenumber $k \sim 10^{-3}$. The shape of the critical modes along the stability curve is illustrated in figure 2, showing contours of constant temperature in a horizontal cross-section. The three states correspond to the circles in figure 1 and depict the transition between the lower and intermediate branches as we consider the new approximation. As we proceed forward along the critical curve the cold fluid progressively penetrates into the warm one and vice versa. The same behaviour is observed when the traditional approximation is used, nevertheless the values of Re_i and G_c required are larger.

As Re_i increases beyond 5×10^5 the new terms in our approximation start becoming important and lead to different behaviour in the linear stability of the system. An analysis of the magnitude of each term in our approximation reveals that the differences observed in figure 1 at high Re_i are due to terms involving the product $v_b u_\theta$, implying the existence of an important centrifugal force acting in azimuthal direction as high rotational speeds are reached. This provides evidence that the traditional formulation, including only the main (radial) contribution of centrifugal buoyancy, is a very good approximation if slow rotation is involved but other contributions may not be neglected in rapidly rotating fluids. Once the critical values given by both approximations differ, we can identify two interesting regions in parameter space. For $Re_i \in [5 \cdot 10^5, 7.7 \cdot 10^5]$ the classical Boussinesq approach yields larger critical G than our approximation, whereas for $Re_i > 7 \cdot 10^5$ the upper branch of the new approximation yields much lower critical values. Moreover, the differences keep increasing as Re_i grows.

The analysis performed reveals the existence of two mechanisms of instability associated with the lower-intermediate and upper branches in figure 1. Different symbols are

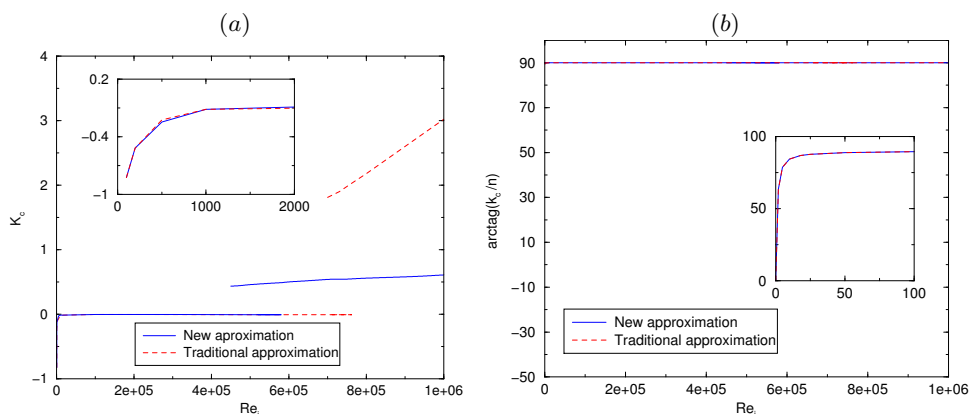


FIGURE 3. (a) Critical axial wavenumber k_c and (b) spiral angle of the modes $\arctan(k_c/n)$ as a function of Re_i for the curves in figure 1. The inset is a close up at low Re_i where the first mechanism stops being dominant and is superseded by spiral modes with angle far from 90 degrees, indeed 0, corresponding to $n = 0$, at $Re_i = 0$.

used to represent the critical values corresponding to each mechanism in each problem. The differences between them are illustrated in figure 3, showing the evolution of the critical axial wavenumber k_c and the angle of the spiral modes $\arctan(k_c/n)$ versus Re_i . Two regions with distinct characteristics are well-defined. The first mechanism of instability has already been presented (see figure 2). Low azimuthal wavenumbers, primarily $n = 1$, and very small axial wavenumbers characterize it. This corresponds to quasi two-dimensional modes and can be readily seen in figure 3(b), showing that the angle of the spiral modes remains constant at about 90 degrees. The inset shows the small initial region where the spiral angle increases progressively until it reaches a vertical position. The second mechanism is characterized by $n > 80$ and $k_c \sim O(1)$, also corresponding to quasi two-dimensional modes (see figure 3b). Another common feature between the two types of instabilities is that the rotational frequency coincides with the angular velocity of the container in both mechanisms and both approximations. This is in agreement with Marezke *et al.* (2013), who have analytically proven that two-dimensional modes with $k = 0$ always rotate at speed A (3.6b) in Taylor–Couette flows without heating. An interesting distinct feature of the second instability mechanism is localization near the inner cylinder. An example of these wall convection modes is shown in figure 4; the critical disturbances are clearly different in the traditional and in our new Boussinesq approximations.

5.2. Differentially rotating cylinders

The traditional approximation for the centrifugal buoyancy neglects the part of the basic flow containing shear, i.e. the B/r term in (3.6b). To quantify the influence of including shear in the centrifugal terms, we perform the same analysis as in the previous section but for differentially rotating cylinders. The amount of shear introduced is characterized

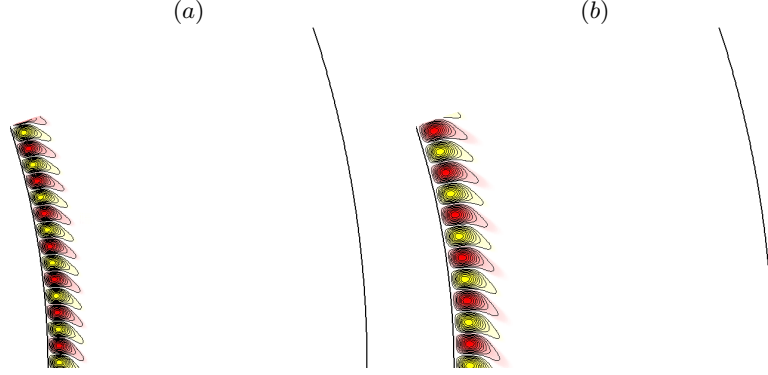


FIGURE 4. Contours of the critical disturbance temperature on a z -constant section for $Re_i = 700000$. (a) Critical mode using the classical Boussinesq approximation. Here $G_c = 489371.47$, $k_c = 1.81$ and $n = 150$. (b) Critical mode using our new approximation. Here $G_c = 207906.92$, $k_c = 0.54$ and $n = 116$. In both cases, only $1/20$ of the domain is shown. There are 10 positive (darker gray; red online) and negative (light gray; yellow online) contours.

by the ratio of angular velocities $\beta = \Omega_i/\Omega_o$; the further β is from unity, the stronger is the shear effect considered.

5.2.1. Weak shear: rotation close to solid body ($\beta = \Omega_i/\Omega_o = 1.006$)

We first consider the case where the container is rotating near solid body. Although shear may be here expected to play only a secondary role, this case serves the purpose of illustrating the importance of including shear effects in the centrifugal term. Figure 5 shows the neutral stability curve for the two approaches considered. Unlike the solid body case, the critical values G_c increase monotonically as Re_i grows. This gives an idea of how important differential rotation is, even if the rotational speeds of each cylinder are only slightly different. Besides shear, centrifugal effects are also important in this configuration. From $Re_i \gtrsim 2 \times 10^5$ on the linear stability curves obtained by using both approximations become quite different. Similar features respect to the solid body case may be identified. At first the traditional approximation gives lower critical value of the Grashof. However, this region is smaller than in the solid-body case and ends at $Re_i \sim 2.9 \times 10^5$ where both curves intersect. From that point on, the stability region predicted by the new approximation is smaller; the differences between the critical values given by both approximations keep increasing as larger Re_i are considered. At the point where both curves first depart from each other Re_i has half the value of that of the solid-body case. Consequently, the rotational speeds for which the new approximation is advisable are quite smaller in the presence of differential rotation.

Critical axial and azimuthal wavenumbers exhibit similar behaviour to the solid-body case and so they are not shown here. Two mechanisms of instability are also found. The first one embraces the region $2 \times 10^5 < Re_i$ and is characterized by $k_c \sim 0$ and $1 \leq n \leq 6$. Modes are similar to those obtained for the first mechanism in the solid body case. A subtle difference can be nevertheless pointed out. In the solid body situation the temperature disturbances fill the whole annulus, whereas differential rotation confines

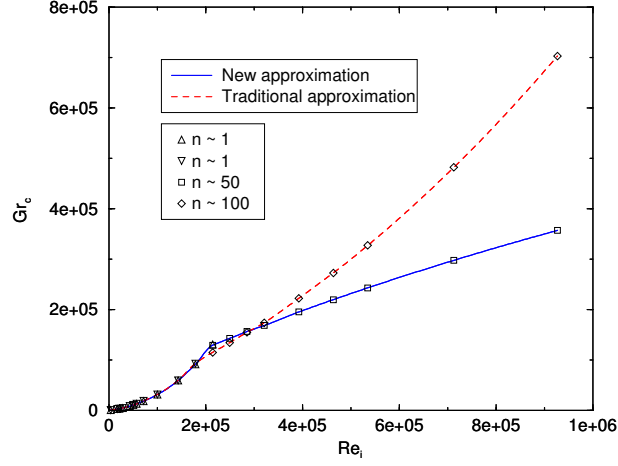


FIGURE 5. Critical Grashof number G_c as function of inner cylinder Reynolds number Re_i for rotation near solid-body ($\beta = 1.006$). Different symbols refer to two distinct instability mechanisms as in figure 1.

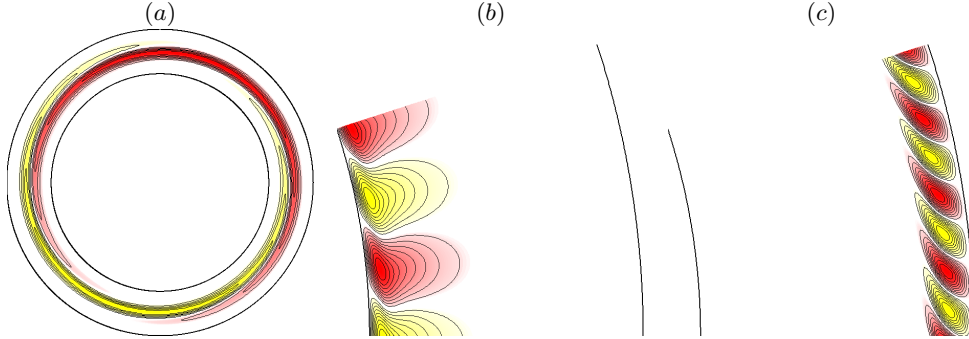


FIGURE 6. Contours of the critical disturbance temperature on a z -constant section. (a) $n = 1$, $Re_i = 178125$, $G_c = 92987.56$. (b)–(c) Comparison of the traditional (b) and new (c) approximation at $Re_i = 285000$ (near the crossover point in figure 5) showing $1/20$ of the annulus. (b) $G_c = 154864.79$, $k_c = 0.24$ and $n = 30$. (c) $G_c = 156547.54$, $k_c = 0.39$ and $n = 75$. There are ten positive (dark gray; red online) and negative (light gray; yellow online) linearly spaced contours.

the perturbation towards the central part (see figure 6a). The second mechanism also presents the same features as in the solid body case, high azimuthal modes and $k_c \in [0.5, 1.5]$, but differences in the flow appear that deserve to be highlighted. In the classical approximation the dominant wall modes are located at the inner cylinder, as it occurs in the solid-body case (figure 6b). In contrast, using the new approximation changes the location of the dominant wall modes to the outer cylinder (figure 6c). In view of these results we can say that considering shear effects in the centrifugal term of the Navier–Stokes equations may be extremely important: not only regarding the linear stability boundary but also the shape and location of the critical modes.

5.2.2. Strong shear: Quasi-keplerian rotation ($\beta = \Omega_i/\Omega_o = 1.58$)

If $1/\eta > \beta > 1$ the angular velocity decreases outward but the angular momentum increases. These flows, known as quasi-Keplerian flows, are used as models to investigate the dynamics and stability of astrophysical accretion disks. Here we choose a typical value $\beta = 1.58$ and as in the previous sections consider a negative temperature gradient in the radial direction, as expected in accretion disks. Figure 7 shows the neutral stability curve for the two approximations considered, as well as entirely neglecting centrifugal effects ($\epsilon = 0$). Here the three curves overlap, implying that shear is the completely dominant mechanism in this regime. As in the weak shear situation studied in the previous section, G_c is also characterized by a monotonous increase as Re_i increases. Surprisingly, shear has a very strong stabilizing effect in this problem: without shear the critical Grashof number is ten times smaller at $Re_i = 10^6$ than in the quasi-Keplerian case.

Depending on the Reynolds number two mechanisms of instability are again found. The first mechanism exhibits a similar flow structure to that observed in the previous case. It also occurs at low Re_i and is localized in the central part of the annulus due to the action of differential rotation. Figure 8(a) shows the countours of the disturbance temperature in an horizontal plane. In contrast to what happens in the weak shear situation, these modes present a clear 3D structure with $k_c \sim -1$. Small azimuthal wavenumbers are involved in this mechanism, ranging from $n = 1$ to $n = 6$. More remarkable differences are found when analysing the second mechanism. High azimuthal modes $n \sim 50$ arise as this mechanism becomes dominant, but unlike the solid-body and weak shear situations, the azimuthal wavenumber decreases as Re_i increases. The same behaviour is observed in the axial wavenumber, so that the spiral angle quickly converges to 90 degrees as observed in the previous sections. Figure 8(b) shows that the instability is characterized by convection wall modes localized at the outer cylinder, as in the case of weak shear using our new approximation. Nevertheless, in quasi-Keplerian flows the dominant modes are always localized at the outer cylinder regardless of how centrifugal terms enter the equations.

6. Summary and discussion

We have identified some weaknesses in how the Boussinesq formulation is typically used to account for centrifugal buoyancy in the Navier–Stokes equations. In particular, this classical approximation neglects the effects associated to differential rotation or strong internal vorticity. This has motivated us to develop a new consistent Boussinesq-type approximation correcting this problem. It consists in keeping the whole density in the advection term of the Navier-Stokes equations and thus it is very easy to implement in an existent solver. The new approximation allows it to accurately treat situations with differential rotation or when strong vortices appear in the interior of the domain, which may cause important centrifugal effects even in flows without global rotation. The latter may be especially relevant in simulations at high Rayleigh numbers (as e.g. in the quest for the ‘ultimate regime’, Ahlers *et al.* 2009). Thus we argue that our formulation for the

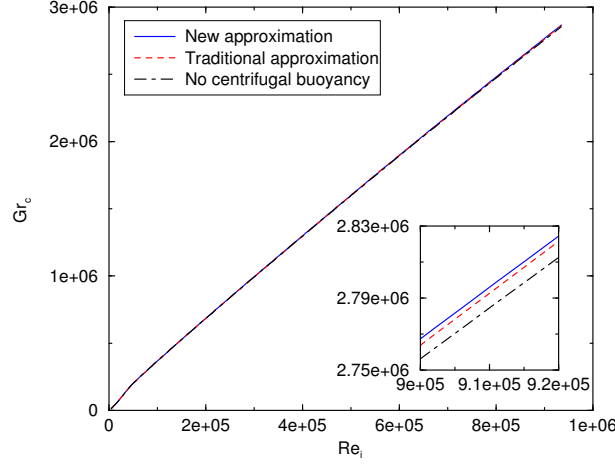


FIGURE 7. Critical Grashof number G_c as function of inner cylinder Reynolds number Re_i for quasi-Keplerian rotation ($\beta = 1.58$). The three curves differ only by about 1% and hence are only distinguishable in the inset (close up).

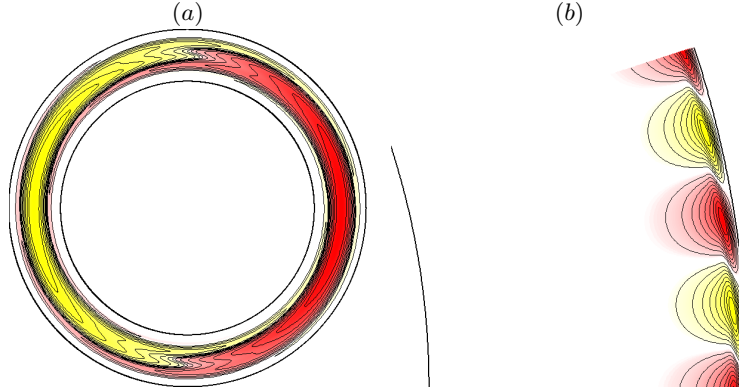


FIGURE 8. Contours of the critical disturbance temperature on a z -constant section. (a) $Re_i = 11681.03$ with $G_c = 4.1268 \times 10^4$, $k_c = -1.05$ and $n = 1$. (b) $Re_i = 584051.72$ with $G_c = 1.8511 \times 10^4$, $k_c = 8.21$ and $n = 38$. Ten positive (dark gray; red online) and negative contours (light gray; yellow online) are displayed. Only 1/20 of the domain is shown in (b).

centrifugal terms should be always implemented whenever the Boussinesq approximation is used.

The relevance of our new approximation has been illustrated with a linear stability analysis of a Taylor–Couette system subjected to a negative radial gradient of temperature. Three different cases have been studied. First, we have considered the container rotating as solid-body, i.e. without differential rotation. The critical values obtained for both traditional and new approximation agree up to $Re_i \sim 5.5 \cdot 10^5$, where discrepancies become significant. Beyond this point the conductive basic flow loses stability to quasi two-dimensional wall modes (aligned with the axis of rotation, as expected from the Taylor–Proudman Theorem) localized at the inner cylinder. Note that the large discrepancy in critical Grashof numbers observed at $Re_i \in [5 \cdot 10^5, 10^6]$ between both

approximations makes it possible to test them against laboratory experiments. For example, in the experiments from Paoletti & Lathrop (2011), the experimental apparatus can easily reach $Re = 10^6$, and the Grashof numbers about 5×10^5 can be obtained with temperature differences about half a degree.

We have also considered the case in which the cylinders rotate at different angular speeds, thus introducing shear. For weak differential rotation shear and centrifugal buoyancy effects compete and the critical values obtained with both approximations differ from each other at lower $Re_i \sim 2 \times 10^5$. Moreover, the new approximation gives rise to wall modes located on the outer cylinder, whereas the traditional approach yields wall modes on the inner cylinder as in the solid-body case. In strongly sheared quasi-Keplerian flows shear is so dominant that centrifugal terms may be entirely neglected in the linear stability analysis (discrepancies in G_c are below 1% regardless of how centrifugal terms enter the equations, if at all). Here the critical modes are always localized at the outer cylinder. Surprisingly, shear has here a very strong stabilizing effect; at least when infinitesimal disturbances are concerned. Note that in more realistic models of accretion disks nonlinear instabilities have been found in similar regimes by Klahr & Bodenheimer (2003). Finally, it is worth noting that testing results with differential rotation in the laboratory is nearly impossible without including axial endwall effects. The large Re_i involved will necessarily trigger instabilities due to the nearly discontinuous angular velocity profile at the junction between axial endwalls and cylinders (Avila 2012).

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